

[03.1 – Materials issues, Fatigue and Damage Tolerance]

## 12.4 \_ Estimation and Testing

<b>Date</b>	27 September 2016 (Tuesday)
<b>Time</b>	14:00–15:30
<b>Place</b>	Track 12 (#209+210+211)
<b>Session Chair: I. Hwang</b>	

<b>12.4.1</b>	<b>14:00–14:30</b>	<b>[2016_0522] DETERMINATION OF DESIGN ALLOWABLES FOR AIRCRAFT COMPOSITE STRUCTURES</b> J.Y. Choung <sup>1</sup> , J.W. Kim <sup>1</sup> , I.H. Hwang <sup>1</sup> ; <sup>1</sup> Korea Aerospace Research Institute, South Korea
<b>12.4.2</b>	<b>14:30–15:00</b>	<b>[2016_0288] A STRESS INTENSITY FACTOR SOLUTION INSPIRED BY SOAP BUBBLES FRAMEWORK</b> R. de Moura Pinho <sup>1</sup> , D. Soria <sup>1</sup> , L. Dimithe Aboumou <sup>1</sup> ; <sup>1</sup> Safran Aircraft Engines, France
<b>12.4.3</b>	<b>15:00–15:30</b>	<b>[2016_0291] ESTIMATION OF TEMPERATURE–DEPENDANT THERMAL CONDUCTIVITY OF MATERIAL WITH GENETIC ALGORITHM</b> Q. Wei–qi <sup>1</sup> , Z. Yu <sup>1</sup> ; <sup>1</sup> CARDC, China